









SoftWind is a 3-year project that aims at developing an experimental setup dedicated to the model testing of floating wind turbines in a wave tank. The experimental modelling methodology is based on an Software-In-the-Loop (SIL) approach that combines numerical modelling with experimental modelling.

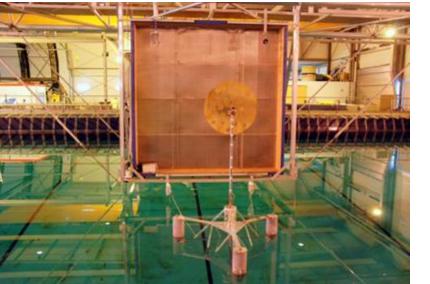
Aerodynamic Thrust Modelling for Wave Tank Tests of FOWT

Rotor Upscaling Thrust Dynamical control/

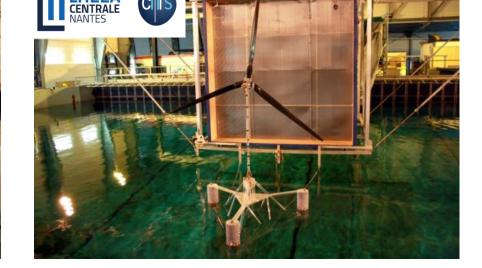
Basic Principle of SIL Methodology for Wave Tank Tests of FOWT

The rotor is represented by an actuator placed at the top of the tower. The actuator is in charge of emulating the response of the rotor. The aerodynamic forces are simulated numerically and are communicated to the ulletactuator. The actuator applies the forces to the model.

	force	loads	command testing	capability (Protor> 5MW)
Froude-scaled « thrust » with drag disk	+ +			-
Froude-scaled thrust geometry (Froude scaling)	+	-	-	-
Rotor performance scaled (Low Reynolds number airfoils)	++	-+	-	-
SIL	++	+/++	++	++



Drag disk





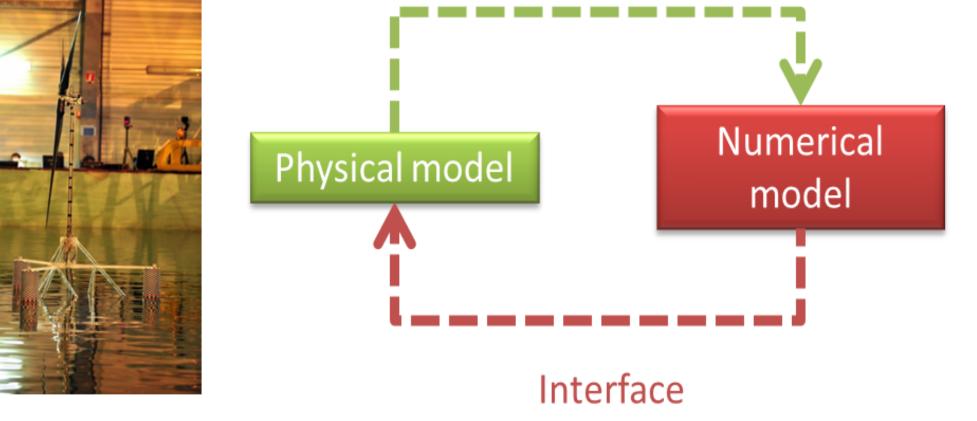


Low Re number airfoils

SIL approach

- The motion of the model is affected by the actuator forces. •
- In response sensors placed on the model inform the numerical model on the nacelle \bullet position and speed.

Interface positions, velocities, accelerations





aerodynamic forces

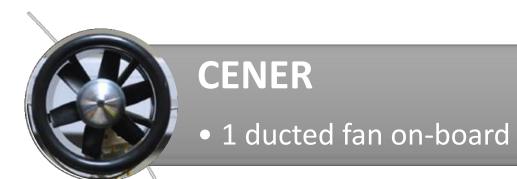
SOFTWIND Current Experimental Setup



A test bench with a single actuator is currently tested. The purpose of this set up is the validation of :

• the communication protocols,

Actuators used in SIL Approaches dedicated to FOWT

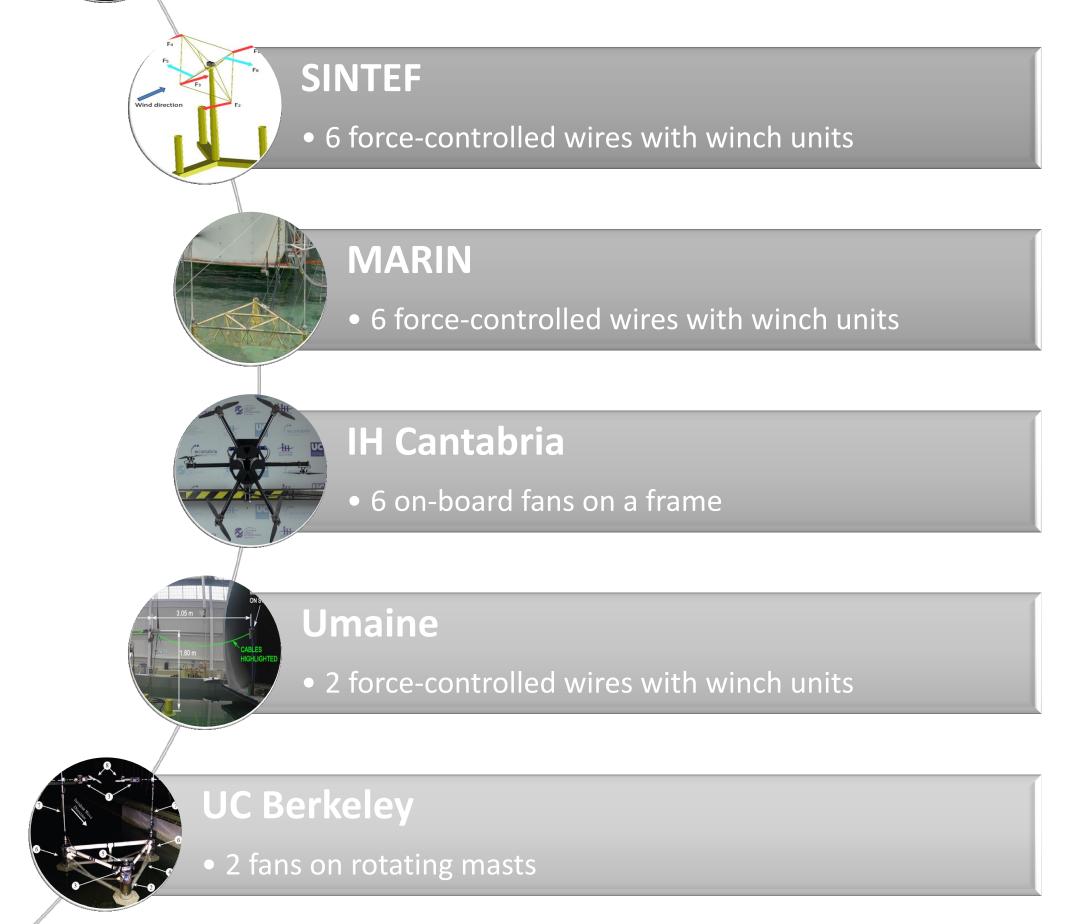




- the real-time execution of the numerical model,
- the motion and force observers,
- the preliminary actuator model identification.

Preliminary validation tests consist in imposed motions by means of a hexapod and will be followed by wave tank tests next September.





- Performance Assessment

• List of wind and waves load cases

Measured and reference force for load case 5 Ξ

× 20

Key milestones of the project





	<i>H_s</i> [m]	T_p [s]	<i>U_w</i> [m/s]	Mean thrust [kN]
LC1	3	5	11.4	680
LC2	4	5	18	320
LC3	6	10	11.4	680
LC4	7	10	18	320
LC5	7	17	11.4	680
LC6	8	17	18	320

• Relative error on the thrust force for the 6 load cases

Load Case	LC1	LC2	LC3	LC4	LC5	LC6
Mean Thrust (N)	25	12.5	25	12	24	12
Error (%)	3	5	5	10	4	5

